DARPA 10-Year Lunar Architecture (LunA-10) Capability Study: Lunar Rail Network Infrastructure Study

#### LSIC Overview

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### **Lunar Rail Network: Project Introduction**

- Challenges with Rovers:
  - Dust, Range, Payload Capacity, Speed, Rolling Resistance, Surface Wear, Recurring Cost
- Potential Solution: Could we build a rail network on the Moon? Should We?
  - Move large amounts of mass easily and efficiently with minimal impact to surface



Notes: Destinations are for illustration only; Not to scale.

Source: Northrop Grumman Lunar Rail Network Study



### LunA-10 & Lunar Rail Mission Space

Mission A: Surface Transport of Lunar-Derived Materials for Cislunar and Earth Orbit ISAM

1) Cost for Space Launch to Low Earth Orbit-Aerospace Security Project (csis.org)

Mission C1:

Lunar Surface

Tourism

2) Commercial Lunar Propellant Architecture: A Collaborative Study of Lunar Propellant Production (isruinfo.com) Projected Value:

(Assuming Starship lowers price of \$1500/kg to LEO<sup>1</sup> to \$300-\$600/kg)

~\$1,400-\$2,600/kg to GEO (4x LEO<sup>2</sup>)

~\$2,000-\$3,800/kg to Moon (>6x LEO<sup>2</sup>)

#### Demand:

- Fuel for Deep Space: 100MT/trip
- Refueling in GEO: 300MT+
- Projected Construction: 100MT's

#### Projected Annual Market Value:

• 500MT-2500MT = \$1.6B-\$8B

#### **Demand:** Driven by price

**Price:** Per one study, ~\$75M per ticket<sup>3</sup> for a tour that includes surface mobility and 20 passengers (price may come down, and buyers would grow)

**Projected Annual Market Value:** 

\$0B to \$1.5B

#### Mission B: Surface Resource Mobility for Lunar Surface Scientific Missions

**Missions:** Lunar Telescopes, Solar Weather Monitoring, Biological Understanding, etc...

**Demand:** Multiple per decade (10's to <200) **Mission Value:** Cost avoidance of bringing power, comm, rovers and equipment. ~10,000 kg per mission = \$38M launch cost plus \$100M+for the equipment  $\rightarrow$  \$150M per mission

#### **Projected Annual Market Value:**

• \$0.5B-\$3B



(newspaceeconomy.ca)

Large mass transport may be key to \$2B to \$13B of annual economic activity by 2035



### **Lunar Rail Network Product Overview**





# **Foundation Forming Equipment Concepts**

- **Surveyor (S):** High-mobility platform with ground penetrating radar and regolith characterization
- **Excavator (E):** Base platform with articulated bucket ladder for • excavation with rippers along chain
- **Hauler (H):** Base platform with coverable bin ۲
- **Compactor (C):** Base platform with a vibratory plate compactor, an articulated vibratory dozer for large rocks and coarse/fine grading, and a coverable bin
- **Assembler (A):** Base platform with robotic arms with for placing, sintering, and welding
- Manager (M): Base platform with outriggers, solar arrays, and communications/PNT

Туре	Approx. Duty Cycle	Notional Unit Launch Mass (kg)	Notional Unit Average Power (kW)
Surveyor	90%	500	0.5
Excavator	70%	1,300	2.3
Hauler	70%	1,400	1.0
Compactor	90%	1,500	0.5
Assembler	70%	1,100	2.1
Manager	100%	1,000	0.1





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Source (All): Northrop Grumman Lunar Rail Network Study



### **Rail Network Design & Analysis**



Train and carrier concept designed with modularity:

- Minimizes launched mass of equipment
- Bogies, frames & transport assembly concepts
  are reconfigurable to payload transport needs

#### Key Design Challenges:

- Lunar Soil: Uncertainty in lunar soil modulus of elasticity for compacted regolith → Potential for deflections >5 cm even with well distributed loads
- Energy: High energy consumption for sintering regolith or extracting and processing metals → Drives goal to minimize processed material for architecture
- Gravity: Reduced gravity vector → reduces traction force and either increases required cant angle in turns or limits speeds in turns



Distribution Statement "A" (Approved for Public Release, Distribution Unlimited) Approved for Public Release: NG24-0751 © 2024 Northrop Grumman Systems Corporation **Lunar Soil Deformation** 



# **Lunar Rail Materials Challenges**

- Limited In-Situ resource availability
  - Metal extraction from regolith is challenging
  - Producible alloys differ from common Terrestrial options
- Extreme temperatures (<-200C to >120C)
  - Raises concerns for ductile to brittle transitions, thermal expansion, and over-aging
- Vacuum environment
  - Lack of surface oxide accelerates wear and cold welding
- High stress between the wheel and rail



Source: Ductile-to-Brittle Transition and Brittle Fracture Stress of Ultrafine-Grained Low-Carbon Steel - PMC (nih.gov)



Source: Apollo Image Archive (asu.edu)



Wear Examples



Source: Cold Welding in Space Mechanisms Due to Fretting (mdpi.com)

Source: Northrop Grumman Lunar Rail Network Study

### These challenges significantly constrain material selection options



### Minimum Viable Experiment (MVE) Phase 1 Objectives

• Goals: Demonstrate commercial ISRU resource extraction framework viability:



• MVE Operations Equipment: 2 Flatbeds (1 Motorized) + 1 Hopper Carrier

### Potential MVE Phase 1 Lunar Rail Activities:

- Site preparation (surveying, excavating & compacting)
- Load hopper with regolith during route construction (demo regolith loading/delivery), deliver regolith to ISRU station & resource processing payloads
- Load engine/flatbed #1 with finished foundry and cast products and deliver to construction fleet (demo goods delivery)
- Use produced rails & sleepers in construction of rail network length
- Install power nodes, demonstrate secondary services
- Place tank on flatbed #2, transport fuel (demo tanker transport) and launch
  - Would bring or harvest a spent propellant tank
  - May need a small propellant tug with launch system to return fuel to orbit



### **MVE to Pilot to Operational Expansion**

- Minimum Viable Experiments (MVE): Prove Lunar surface viability and refines architecture design
- Pilot Expansion: Uses second generation carriers/engines & construction equipment (same track gauge)
- Operational Rail Network: Settlement 1 infrastructure complete, fully operational and generating revenue

#### **Concept of Expansion**

MVE Landing Site & Initial Processing Facility



#### Lunar Rail Network would Scale as Lunar Surface Utilization Grows

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Elevation<sup>1</sup>

# **MVE Site Selection & Expansion**

Elevation and Regional Context<sup>1</sup>



Lunar Rail Network Study: Design Reference Plan 1



Illumination & Slope<sup>2</sup>



Lunar Rail Network Study: Design Reference Plan 2



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Sources: 1) (Adapted from)

- Barker, M.K., et al. (2020), Improved LOLA Elevation Maps for South Pole Landing Sites: Error Estimates and Their Impact on Illumination Conditions, Planetary & Space Science, in press, doi:10.1016/j.pss.2020.105119.
- Stopar J. and Meyer H. (2019) Annual Illumination and Topographic Slope of the Moon's South Polar Ridge, Lunar and Planetary Institute Regional Planetary Image Facility, LPI Contribution

2179, https://repository.hou.usra.edu/handle/20.500.11753/1264



# **Investment & Trip Price by Phase Summary**

# Non-Recurring Engineering & Operations

- Concept Development and Design of Infrastructure and Transport System
  - Train, Civil Structures, Substations, Construction Equipment
- Launch of material to Moon Base
- Test track and equipment/system maturation
- Lunar Rail staged infrastructure buildup
- Manufacturing, integration, test and launch of construction equipment and Earth-sourced components/assemblies

# Recurring Engineering & Operations

- Operational cost of train (energy, comms/PNT, loading/unloading)
- ISRU-derived manufacturing of new elements and maintenance of existing elements (Railcars, beds, etc.)

Source: Northrop Grumman Lunar Rail Network Study

Parameter	Phase 1	Phase 2	Phase 3	Phase 4	Phase 5
Average Trip Length [km]	0.2	2.4	3.9	24	128
Average Trip Payload [kg]	200	5,000	32,000	45,000	83,000
Total Launched Mass by Phase [kg]	16,100	25,300	64,200	90,000	167,000
ISRU Metallic Mass Required [kg]	1,800	43,300	212,000	520,000	3,700,000
TOTAL Estimated NRE [\$B]	0.5-1.0	0.5-1.0	2.0-3.0	2.0-3.0	4.0-5.0
TOTAL Estimated RE [\$B]		<0.1	0.1-0.2	0.3-0.5	2.0-3.0
TOTAL Goal Avg Price per Trip [\$M]	N/A	8.0	8.0	5.0	2.5
Mass Price Efficiency [\$USD / kg-km]	17,700,000	660	51	3.4	0.23

**Note:** Numbers shown are to ballpark costs and do not include contingency. At the final system concept review the numbers will be updated along with the assumptions below.

#### Pilot Lunar Rail – Key Architecture Assumptions

• Max Transport Mass: 100,000 kg – 125,000 kg

Phase 2+ Price Per Trip: ~\$2.5M-\$8M

- Cost Assumptions:
  - Development Cost \$30,000 per kg<sup>1</sup> (Equipment, Detailed Components)
  - Development Cost \$3,000 per kg (Maintenance Depot Building)
  - Cost of Launch to Moon Base \$10,000 per kg (~7x Falcon 9 to LEO)<sup>2</sup>
  - Cost of Manufacturing Metal Parts on the Moon \$500 per kg
  - Cost of Manufacturing Equipment on Earth \$5,000 per kg
  - Cost of Energy on the Moon \$100 per kWh
  - Maintenance: 5% per year of manufacturing cost for infrastructure (20-year life), 20% per year for carriers (5-year life)

Additional Sources: 1) Demand Drivers of the Lunar and Cislunar Economy (newspaceeconomy.ca), Page 123 2) Cost for Space Launch to Low Earth Orbit- Aerospace Security Project (csis.org)

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### Cross-Mass Mobility: Trip Momentum Cost Efficiency White Space Plot Draft Result

Note: Derivation of the terrestrial transportation and rover values will be discussed in the LunA-10 Final Report



Initial Take-Aways:

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- Rover constrained by capacity
- o Lunar rail constrained by infrastructure cost
- Slope is steeper with higher infrastructure cost, offset by increasing capacity of transport
- Refinement of total costs and capacities is important to understanding rover vehicle versus rail overlap region
- Developing projected transport service demand is important to determine when rail makes strict economic sense with all other things being equal

Cross-Mass Mobility: Momentum Cost Efficiency					
Transport Methodology	X Axis: kg*km/min	Y-Axis: \$/kg-km			
Lunar Rover	42	109			
Lunar Truck on Road	1250	24.7			
Next Gen Lunar Rover	333	2.13			
Phase 2 Lunar Rail Transport	830	656			
Phase 3 Lunar Rail Transport	5400	63.6			
Phase 4 Lunar Rail Transport	23000	4.65			
Phase 5 Lunar Rail Transport	42000	0.23			
Beyond Phase 5 Lunar Rail Network	200000	0.05			
Cableway (Asmara-Massawa)	35900	0.000612			
US Route 70 Flatbed	66000	0.000045			
US Freight Rail Transport (Long)	24500000	0.000016			
US Freight Rail Transport (Short)	6250000	0.000020			

#### There is a rover to rail cross-over point around >1000 kg-km/min and/or achieving <\$5-10/kg-km

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