



## ANTARES™ Medium-Class Space Launch Vehicle

Designed to provide responsive and low-cost access to space, Antares is a two-stage vehicle (with optional third stage) that provides low-Earth orbit (LEO) launch capability for payloads weighing up to 8,000 kg. Developed by Northrop Grumman, Antares completed a risk reduction mission and a demonstration of commercial re-supply services for the International Space Station (ISS) under a NASA Commercial Orbital Transportation Services (COTS) agreement in 2013. Northrop Grumman commenced delivery of cargo to the International Space Station under the NASA Commercial Resupply Services 1 (CRS1) contract from 2014 to 2019. In early 2016, Northrop Grumman was selected to continue cargo deliveries

under the CRS2 contract. Missions under CRS2 commenced in 2019 and will continue until at least 2024.

The Antares launch system utilizes Northrop Grumman's proven MACH avionics system and many management approaches, engineering standards, production and test processes common to Northrop Grumman's family of successful small-class Pegasus® and Minotaur launch vehicles. The Antares design features RD-181 first stage engines that provide robust performance and excellent reliability. The second stage features Northrop Grumman designed and built elements, including the CASTOR® 30XL solid rocket motor, advanced composite structures, including the payload fairing, and MACH avionics that are incorporated into all Northrop Grumman Space Launch Vehicles.

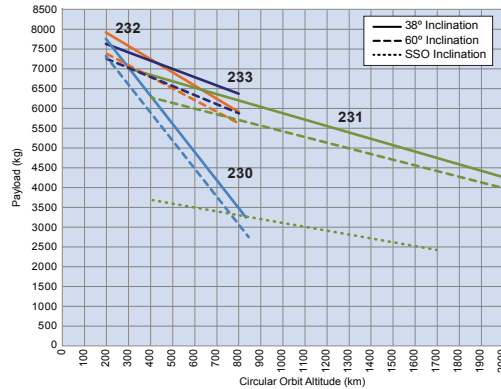
### KEY FEATURES

- Flight-proven launch vehicle
- Provides substantial payload performance into a variety of low inclination low-Earth and sun-synchronous orbits and interplanetary trajectories
- Streamlined vehicle/payload integration and testing via standardized interfaces reduce time from encapsulation to lift-off
- 3.9 meter fairing
- Capable of launching single and multiple payloads
- Launching from Wallops Flight Facility (WFF), Virginia
- Up to 8,000 kg to low-Earth orbit

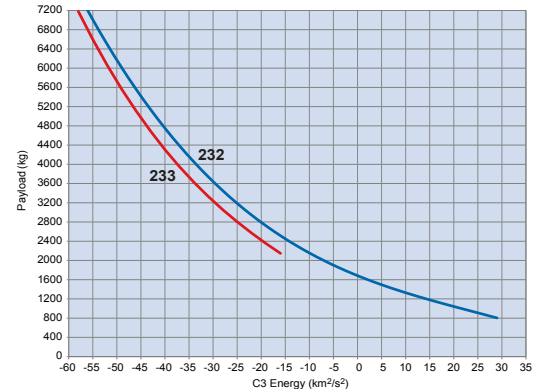
# ANTARES™

## PERFORMANCE

Antares Performance to Circular Orbits (WFF)



Antares Performance to 38° High Energy Orbits (WFF)



## PAYLOAD FAIRING

- Diameter: 3.9 m
- Height: 9.9 m
- Structure: Honeycomb core, composite face sheets

## STAGE 1

- Two NPO Energomash RD-181 engines with independent thrust vectoring
- Liquid oxygen/kerosene fueled
- Northrop Grumman responsible for system development and integration
- Core tank design and design verification by KB Yuzhnoye (Zenit-derived)
- Core tank production by Yuzhmash
- Avionics stage controller uses flight-proven Northrop Grumman MACH components

## STAGE 2

- Northrop Grumman CASTOR® 30XL solid motor with thrust vectoring
- MACH avionics

## MORE INFORMATION

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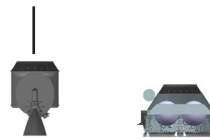
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## ANTARES CONFIGURATION NUMBERING

First Stage	Second Stage	Third Stage
2 – Two NPO Energomash RD-181 LOX/Kerosene Fueled Engines	3 – CASTOR® 30XL Solid Motor	0 – None
		1 – Orbital Adjust Module (OAM)
		2 – STAR™ 48-Based Third Stage
		3 – Orion 38-Based Third Stage

## STAGE 3

### Optional STAR™ 48BV-Based Third Stage



### Optional Orbit Adjust Module Third Stage (OAM)

- Monopropellant Hydrazine System
- Helium pressure regulated (Northrop Grumman GEOStar Bus and HAPS Heritage)

- Optional Orion 38-Based Third Stage
- Northrop Grumman Solid Rocket Motor Upper Stages
- Thrust vector guidance and control
- 3-axis stabilized satellite orbit insertion

## MISSION PARTNERS

### Northrop Grumman

Prime integrator, systems engineering, avionics, primary structure, testing and software. Overall Stage 1 development and integration, Stage 2 motor, composite structures

### KB Yuzhnoye/Yuzhmash

Stage 1 core design, production and verification

### NPO Energomash

Stage 1 engines

